Extension of the MOOD Method to Eulerian Simulations of Multi-Material Compressible Flows on Unstructured Meshes

2D preliminary results with a 6-equation model and a fourth-order method

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OUTLINE

Overview of the MOOD method

A 6-equation model for multi-material flows

An Eulerian discretization on unstructured meshes

Adaptation of the MOOD detection criteria

Preliminary numerical results



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To date

- Developed during my Ph.D. in Toulouse (FR) under S. Clain & R. Loubère
- Very-high-order Finite Volume method for single-material Euler equations
- Alternative to WENO limiting on multidimensional unstructured meshes
- Successfully tested up to 6^{th} -order of accuracy on 3D polyhedral meshes
- Papers: JCP 2011, CAF 2012, IJNMF 2013 \implies public.lanl.gov/diot

On-going

- Extension to multi-material compressible flows (S.Diot, LANL.)
- Efficient OpenMP parallelization of 3D code (G.Moebs, Nantes, FR.)
- MOOD concept for higher-order remapping (R.Loubère, Toulouse, FR.)
- Steady-state problems, shallow water, diffusion (S.Clain, Guimaraes, PT.)
- Extension with ADER one-step space-time discr. (M.Dumbser, Trento, IT.)





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 $\underline{Main \ ideas}$

- Use only one unlimited polynomial reconstruction per cell
- Check after time update (a posteriori) if the solution is acceptable
- If not, locally decrement the cell polynomial degree and recompute
- In the worst case, use the first-order scheme as parachute

Tools

- Robust and effective framework developed in JCP 2011
- Need for Detection Criteria to define an *acceptable solution*!



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Reference: J. Comput. Phys. 228 (2009) 1678-1712. (SPB)

The non-conservative hyperbolic system is given by

$\int \partial_t U$	+	$\nabla \cdot F(U)$	=	0
$\partial_t(lpha_1)$		$\mathbf{V} \cdot abla(lpha_1)$		
$\partial_t(\alpha_1\rho_1e_1)$		$\nabla \cdot (\alpha_1 \rho_1 e_1 \mathbf{V}) + \alpha_1 p_1 \nabla \cdot \mathbf{V}$		
$\partial_t(\alpha_2\rho_2e_2)$		$\nabla \cdot (\alpha_2 \rho_2 e_2 \mathbf{V}) + \alpha_2 p_2 \nabla \cdot \mathbf{V}$		

• Conservation of material mass, momentum and total energy

$$U = \begin{pmatrix} \alpha_1 \rho_1 \\ \alpha_2 \rho_2 \\ \rho \mathbf{V} \\ \rho E \end{pmatrix} \quad \text{and} \quad F(U) = \begin{pmatrix} \alpha_1 \rho_1 \mathbf{V} \\ \alpha_2 \rho_2 \mathbf{V} \\ \rho \mathbf{V} \otimes \mathbf{V} + pI \\ (\rho E + p) \mathbf{V} \end{pmatrix}$$



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- Conservation of material mass, momentum and total energy
- Non-conservative volume fraction advection equation + relaxation terms
- Non-conservative material internal energies equations + relaxation terms

$$\sum_{k} \alpha_{k} = 1, \quad \rho = \sum_{k} \alpha_{k} \rho_{k}, \quad p = \sum_{k} \alpha_{k} p_{k}, \quad \rho c^{2} = \sum_{k} \alpha_{k} \rho_{k} c_{k}^{2} \text{ and } \sum_{k} \alpha_{k} \rho_{k} e_{k} \stackrel{!}{=} \rho e_{k}$$

$$\longrightarrow \text{ Solved in 3 steps: 1) Solve the system w/o relaxation terms 2) Perform the relaxation step 3) Correct the material internal energies
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NOTATION

We consider a polygonal mesh of domain $\Omega \subset \mathbb{R}^2$.

Geometry

Index sets





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FIRST-ORDER DISCRETIZATION

 $Conservative \ part \ of \ the \ system$

$$\begin{split} \int_{\Omega_i} \partial_t U(\mathbf{x},t) \, d\mathbf{x} &= -\int_{\Omega_i} \nabla \cdot F(U(\mathbf{x},t)) \, d\mathbf{x}, \\ &= -\sum_{j \in \underline{\nu}(i)} \int_{f_{ij}} F(U(\mathbf{x},t)) \cdot \mathbf{n}_{ij} \, dS, \end{split}$$

leads to

$$U_i^{n+1} = U_i^n - \frac{\Delta t}{|\Omega_i|} \sum_{j \in \underline{\nu}(i)} |f_{ij}| \mathbb{F}(U_i, U_j, \mathbf{n}_{ij}),$$

where \mathbb{F} is an approximate Riemann Solver in the \mathbf{n}_{ij} direction.

 $\implies \text{We use HLLC (see SPB)}$ (which also provides the normal velocity $u^{\star}_{\mathbf{n}_{ij}} = \mathbf{V}^{\star} \cdot \mathbf{n}_{ij}$)



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FIRST-ORDER DISCRETIZATION

Volume fraction equations

Using that

$$\partial_t \alpha_1 + \mathbf{V} \cdot \nabla(\alpha_1) = \partial_t \alpha_1 + \nabla \cdot (\alpha_1 \mathbf{V}) - \alpha_1 \nabla \cdot \mathbf{V},$$

we have

$$\begin{split} \int_{\Omega_i} \partial_t \alpha_1 \, d\mathbf{x} &= -\int_{\Omega_i} \left[\nabla \cdot (\alpha_1 \mathbf{V}) - \alpha_1 \nabla \cdot \mathbf{V} \right] d\mathbf{x}, \\ &\stackrel{approx}{\approx} &- \sum_{j \in \underline{\nu}(i)} \left[\int_{f_{ij}} (\alpha_1 \mathbf{V}) \cdot \mathbf{n}_{ij} \, dS - \alpha_1 \int_{f_{ij}} \mathbf{V} \cdot \mathbf{n}_{ij} \, dS \right], \end{split}$$

and

$$(\alpha_1)_i^{n+1} = (\alpha_1)_i^n - \frac{\Delta t}{|\Omega_i|} \sum_{j \in \underline{\nu}(i)} |f_{ij}| \left[\alpha_1^{\star} u_{\mathbf{n}_{ij}}^{\star} - (\alpha_1)_i^n (u_{\mathbf{n}_{ij}}^{\star}) \right],$$

where $\begin{cases} u^{\star}_{\mathbf{n}_{ij}} \text{ is given by the Riemann Solver} \\ \alpha^{\star}_{1} \text{ is obtained as constant along fluid trajectories} \end{cases}$





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FIRST-ORDER DISCRETIZATION

Energies equations

Following the same track, we get

$$\begin{split} \int_{\Omega_i} \partial_t (\alpha_k \rho_k e_k) \, d\mathbf{x} &= -\int_{\Omega_i} \left[\nabla \cdot \left((\alpha_k \rho_k e_k) \mathbf{V} \right) - (\alpha_k p_k) \nabla \cdot \mathbf{V} \right] \, d\mathbf{x}, \\ &\stackrel{approx.}{\approx} - \sum_{j \in \underline{\nu}(i)} \left[\int_{f_{ij}} ((\alpha_k \rho_k e_k) \mathbf{V}) \cdot \mathbf{n}_{ij} \, dS - (\alpha_k p_k) \int_{f_{ij}} \mathbf{V} \cdot \mathbf{n}_{ij} \, dS \right], \end{split}$$

and

$$(\alpha_k \rho_k e_k)_i^{n+1} = (\alpha_k \rho_k e_k)_i^n - \frac{\Delta t}{|\Omega_i|} \sum_{j \in \underline{\nu}(i)} |f_{ij}| \left[(\alpha_k \rho_k e_k)^* u_{\mathbf{n}_{ij}}^* - (\alpha_k p_k)_i^n (u_{\mathbf{n}_{ij}}^*) \right],$$

where $\begin{cases} u_{\mathbf{n}_{ij}}^{\star} \text{ is given by the Riemann Solver} \\ (\alpha_k \rho_k e_k)^{\star} \text{ is determined by Hugoniot relation proposed in SPB} \end{cases}$



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PRESSURE RELAXATION - ENERGIES CORRECTION

 $\underline{Stiffened \ gas \ EOS} \quad p_k = \rho_k e_k (\gamma_k - 1) - \Pi_k \gamma_k$

<u>Relaxation step</u> (only affects α_k)

• After manipulations, the relaxation step consists in solving (for p)

$$\sum_{k} (\alpha_k \rho_k) \nu_k(p) = 1, \ (\alpha_k \rho_k) \ being \ constant,$$

where the specific volume is

$$\nu_{k}(p) = \nu_{k}^{0} \frac{p_{k}^{0} + \gamma_{k} \pi_{k} + (\gamma_{k} - 1)\hat{p}_{I}}{p + \gamma_{k} \pi_{k} + (\gamma_{k} - 1)\hat{p}_{I}}$$

- Use Newton's iterative method to find the relaxed pressure p
- Compute $\rho_k = \nu_k(p)^{-1}$ and deduce the corrected vol. fraction $\alpha_k = \frac{\alpha_k \rho_k}{\rho_k}$

Energies correction (only affects $\alpha_k \rho_k e_k$)

- Compute the mixture pressure $p_{mix} = \left(\rho e \sum_{k} \frac{\alpha_k \gamma_k \pi_k}{\gamma_k 1}\right) / \left(\sum_{k} \frac{\alpha_k}{\gamma_k 1}\right)$
- Deduce the corrected material energies from EOS $e_k = e_k(\alpha_k, \alpha_k \rho_k, p_{mix})$





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$In \ space$

- Integrals on face f_{ij} are approximated by quadrature rules (r=1,...,R)
 - ★ Points $q_{ij,r}$ on face f_{ij}
 - * Weights $\xi_r \ge 0$ such as $\sum_r \xi_r = 1$

$$\int_{f_{ij}} F(U) \cdot \mathbf{n}_{ij} dS \stackrel{LO}{\approx} \mathbb{F}(U_i, U_j, \mathbf{n}_{ij}) \stackrel{HO}{\approx} \sum_{r=1}^{R} \xi_r \mathbb{F}(U_{ij,r}, U_{ji,r}, \mathbf{n}_{ij})$$

Illustration:





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- Higher-order approximations
 - $\star\,$ Compute (LS) polynomial reconstructions $\widetilde{U}_i(\mathbf{x})$ on every Ω_i
 - * Evaluate reconstructions at $q_{ij,r}$: $U_{ij,r} = \widetilde{U}_i(q_{ij,r}) / U_{ji,r} = \widetilde{U}_j(q_{ij,r})$

 \Rightarrow Same method for face integrals in non-conservative equations

<u>In time</u>

- We use the RK3 TVD method (3 sub-steps)
- The MOOD method is performed over each substep



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Detection Criteria

- **PAD**: Physical Admissibility Detection
 - $\star\,$ Minimal conditions to ensure physicality of the solution
 - $\star\,$ For the single-material case: $\rho>0,\,p>0,\,e>0,\,etc.$
- **DMP**: Discrete Maximum Principle
 - \star Designed to prevent spurious oscillations from appearing

Check if u_i^* fulfills the Discrete Maximum Principle: $\min_j(u_i^n, u_j^n) \le u_i^* \le \max_j(u_i^n, u_j^n)$

- $\star\,$ DMP violation must be allowed at smooth extrema to reach VHO
- **u2**: Smoothness Detection
 - \star Only checked if the DMP is violated, $i.e.~2^{nd}$ filter
 - $\star\,$ Distinguishes a discontinuity from a smooth extremum
 - \star Uses approximations of *curvatures* over a local neighborhood



 \Rightarrow For the single-material case: DMP+u2 applied on density!



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MOOD CRITERIA - TO THE MULTI-MATERIAL ONE

Detection Criteria

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 - * Positivity of material densities: $\alpha_k \rho_k > 0, \ \forall k$,
 - * Positivity of material energies: $\alpha_k \rho_k e_k > 0, \ \forall k$,
 - * Positivity of mixture pressure: $p = \sum_k \alpha_k p_k > 0.$

• **DMP+u2** applied on mixture density $\rho = \sum_k \alpha_k \rho_k$ <u>Remarks</u>

- PAD seems to ensure robustness, but still to be proven
- DMP+u2 on ρ reduces the appearance of spurious oscillations
- Preliminary choice \longrightarrow May be improved in the future



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NUMERICAL RESULTS - GENERAL COMMENTS

About the MUSCL-like implementation

- 1D results are computed with my research code using SPB model
- 2D results are computed with LANL xRage code (dim. splitting)
 - \star Modified 6-eq. model of Miller-Puckett (Francois et~al, CaF 2012)
 - $\star\,$ Interface Preserver is always enabled to reduce interface diffusion
 - $\star\,$ Only cartesian meshes without AMR technique

About the MOOD implementation

- 2D unstructured (polygonal) multi-material research code
- Polynomial reconstructions of arbitrary degree, here \mathbb{P}_1 & \mathbb{P}_3
- MOOD parameters are set independently of cases/meshes

Numerical parameters

- The CFL coefficient is set to 0.6
- The smallest volume fraction is always 10^{-8}





Abgrall shock tube - 800 cells - Solution



$t_{end} = 0.01s$	ρ	u	р	γ	π	Boundary Cond.
Left State on $[0.0; 0.5]$	1	0	500	1.4	0	Reflective at x=0.0
Right State on $[0.5; 1.0]$	1	0	0.2	1.6	0	Reflective at x=1.0



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The MOOD Method

Abgrall shock tube - 800 cells - Solution



$t_{end} = 0.01s$	3	ρ	u	р	γ	π	Boundary Cond.
Left State on [0.0	; 0.5]	1	0	500	1.4	0	Reflective at x=0.0
Right State on [0.5	5; 1.0]	1	0	0.2	1.6	0	Reflective at x=1.0



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The MOOD Method

6-eq. Model

Discretization

MOOD Criteria

Abgrall shock tube - 800 cells - Convergence



WATER-AIR SHOCK TUBE - 1000 CELLS - SOLUTION



$t_{end}=220\mu s$	ρ	u	р	γ	π	Boundary Cond.
Left State on $[0.0; 0.7]$	1000	0	1.10^{9}	4.4	6.10^{8}	Reflective at x=0.0
Right State on [0.7; 1.0]	50	0	1.10^{5}	1.4	0	Reflective at x=1.0



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WATER-AIR SHOCK TUBE - 1000 CELLS - SOLUTION

Effect of the energy correction:

- SPB: Energy correction from mixture pressure as presented before
- **RED**: Simple REDistribution algorithm
 - $\star\,$ Compute internal energy delta: $\delta_{\rho e} = \rho e \sum_k (\alpha_k \rho_k e_k)^{old}$

* Distribute it according to
$$\frac{(\alpha_k \rho_k)}{\rho}$$
: $(\alpha_k \rho_k e_k)^{new} = (\alpha_k \rho_k e_k)^{old} + \frac{(\alpha_k \rho_k)}{\rho} \delta_{\rho e}$

L1 Errors:

	MUSCL	MOC	D-P1	MOOD-P3		
	SPB	SPB	RED	SPB	RED	
Density	3.466E-03	3.271E-03	2.246E-03	2.595E-03	1.716E-03	
Velocity	8.028E-03	8.499E-03	3.276E-03	6.625E-03	3.233E-03	
Pressure	4.156E-03	3.971E-03	3.937E-03	3.734E-03	3.236E-03	
Sp.Int.En.	4.932E-03	3.947E-03	2.720E-03	2.795E-03	2.131E-03	

 \longrightarrow Energy correction significantly affects the solution

 \implies Need more work



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WATER-AIR SHOCK TUBE - 1000 CELLS - SOLUTION



$t_{end}=220\mu s$	ρ	u	р	γ	π	Boundary Cond.
Left State on $[0.0; 0.7]$	1000	0	1.10^{9}	4.4	6.10^{8}	Reflective at x=0.0
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TRIPLE POINT PROBLEM - INITIALIZATION

• Galera *et al.* "A two-dimensional unstructured cell-centered multi-material ALE scheme using VOF interface reconstruction"



- Final time = 5.0
- Wall boundary conditions



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TRIPLE POINT - 2^{nd} -Order comparison

MOOD-P1 1120x480 Density

xRAGE 1120x480 Density

$\longrightarrow 2^{nd}$ -order results are qualitatively equivalent (validation)



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TRIPLE POINT - 4^{th} -order vs 2^{nd} -order

MOOD-P3 560x240 Density

xRAGE 1680x720 Density

 $\rightarrow 4^{th}$ -order method reveals small structures on a 9× coarser mesh! (~13.5× faster)

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TRIPLE POINT - INTERFACE DIFFUSION COMPARISON

MOOD-P3 1120x480 Vol. Fraction

xRAGE 1120x480 Vol. Fraction

 \longrightarrow Limiting interface diffusion in a 2^{nd} -order method does not have the same effect than increasing the method order of accuracy.



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$\operatorname{R22}\,\operatorname{Bubble}$ - Initialization

Experiment

• J.F.Haas, B.Sturtevant. "Interaction of weak shock waves with cylindrical and spherical gas inhomogeneities"

Simulations

- J.J.Quirk, S.Karni. "On the dynamics of a shock-bubble interaction"
- S.Kokh, F.Lagoutière. "An anti-diffusive numerical scheme for the simulation of interfaces between compressible fluids by means of a five-equation model"



- Final time = $1020.0 \ \mu s$
- Wall boundary conditions on top/bottom



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R22 BUBBLE - 2^{nd} -Order comparison

MOOD-P1 1600x320 Density

xRAGE 1600x320 Density

$\rightarrow 2^{nd}$ -order results are qualitatively equivalent (validation)



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R22 BUBBLE - 4^{th} -Order vs 2^{nd} -Order

MOOD-P3 1200x240 Vol. Fraction

xRAGE 3200x640 Vol. Fraction

 $\longrightarrow 4^{th} \text{-order}$ method reveals small structures on a 7× coarser mesh! (~9.5× faster)



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R22 Bubble - Structured vs Unstructured

 $\begin{array}{c} \mbox{MOOD-P3}\\ \mbox{Unstructured}\\ \mbox{101923 cells}\\ \mbox{$h > 5.0 \times 10^{-4}$}\\ \mbox{Density} \end{array}$

 $\begin{array}{c} \mbox{MOOD-P3} \\ \mbox{Structured} \\ \mbox{288000 cells} \\ \mbox{$h = 3.7 \times 10^{-4}$} \\ \mbox{Density} \end{array}$

 \longrightarrow The ability to reveal structures is *not affected* by unstructured meshes



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CONCLUSION & FUTURE WORK

Conclusion

- First extension of the MOOD method to a 6-eq. multi-material model
- Developed for unstructured meshes with diffuse interface treatment
- Better efficiency of MOOD- \mathbb{P}_3 (4th-order) vs MUSCL (2nd-order)
- MOOD- \mathbb{P}_3 reveals the small structures *earlier* (space/time)

Future work

- Develop a 2D axisymmetric version of the research code
- Quantitatively assess the relevancy of the 4th-order method
- Demonstrate the positivity-preserving property of the MOOD method
- Develop a MOOD direct Eulerian VOF method on unstructured meshes



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THANKS FOR YOUR ATTENTION!

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http://public.lanl.gov/diot



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EXTRA SLIDES



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Form of polynomial reconstruction of degree d

$$\widetilde{U}_{i}(\mathbf{x}) = U_{i}^{n} + \sum_{1 \le |\boldsymbol{\alpha}| \le d} \mathcal{R}_{\boldsymbol{\alpha}} \left((\mathbf{x} - \mathbf{c}_{i})^{\boldsymbol{\alpha}} - \frac{1}{|K_{i}|} \int_{K_{i}} (\mathbf{x} - \mathbf{c}_{i})^{\boldsymbol{\alpha}} d\mathbf{x} \right)$$

• $\boldsymbol{\alpha} \in \mathbb{R}^m$ multiindex \Longrightarrow covers all monomials

• $\mathcal{R}_{\alpha} \in \mathbb{R}$ unknown poly. coefficients s.t. $\#(\mathcal{R}) = \frac{\prod_{i=1}^{m} (d+i)}{m!} - 1$

Computation of \mathcal{R}_{α}

- Approximations of mean values on a local stencil
- Local stencil contains more than $\#(\mathcal{R})$ cells
- Resolution of an overdetermined linear system \equiv Least-Squares



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Form of polynomial reconstruction of degree d

$$\widetilde{U}_{i}(\mathbf{x}) = U_{i}^{n} + \sum_{1 \le |\boldsymbol{\alpha}| \le d} \mathcal{R}_{\boldsymbol{\alpha}} \left((\mathbf{x} - \mathbf{c}_{i})^{\boldsymbol{\alpha}} - \frac{1}{|K_{i}|} \int_{K_{i}} (\mathbf{x} - \mathbf{c}_{i})^{\boldsymbol{\alpha}} d\mathbf{x} \right)$$

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Computation of \mathcal{R}_{α}

- Approximations of mean values on a local stencil
- Local stencil contains more than $\#(\mathcal{R})$ cells
- Resolution of an overdetermined linear system \equiv Least-Squares





$$\begin{array}{l} \text{Mean value of } \widetilde{U}_i^n \text{ on a } K_j \\ U_i^n + &\sum_{1 \le |\boldsymbol{\alpha}| \le d} \mathcal{R}_{\boldsymbol{\alpha}} \left(\frac{1}{|K_j|} \int_{K_j} (\mathbf{x} - \mathbf{c_i})^{\boldsymbol{\alpha}} d\mathbf{x} - \frac{1}{|K_i|} \int_{K_i} (\mathbf{x} - \mathbf{c_i})^{\boldsymbol{\alpha}} d\mathbf{x} \right) \end{array}$$

Packing in a Matrix-vector form

$$\begin{pmatrix} \bar{\mathbf{x}}_{1}^{(1,0,0)} & \bar{\mathbf{x}}_{1}^{(0,1,0)} & \bar{\mathbf{x}}_{1}^{(0,0,1)} & \cdots & \bar{\mathbf{x}}_{1}^{(0,0,d)} \\ \bar{\mathbf{x}}_{2}^{(1,0,0)} & \bar{\mathbf{x}}_{2}^{(0,1,0)} & \bar{\mathbf{x}}_{2}^{(0,0,1)} & \cdots & \bar{\mathbf{x}}_{2}^{(0,0,d)} \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ \bar{\mathbf{x}}_{N}^{(1,0,0)} & \bar{\mathbf{x}}_{N}^{(0,1,0)} & \bar{\mathbf{x}}_{N}^{(0,0,1)} & \cdots & \bar{\mathbf{x}}_{N}^{(0,0,d)} \end{pmatrix} \begin{pmatrix} \mathcal{R}_{(1,0,0)} \\ \mathcal{R}_{(0,1,0)} \\ \mathcal{R}_{(0,0,1)} \\ \vdots \\ \mathcal{R}_{(0,0,d)} \end{pmatrix} = \begin{pmatrix} U_{1}^{n} - U_{i}^{n} \\ U_{2}^{n} - U_{i}^{n} \\ \vdots \\ U_{N}^{n} - U_{i}^{n} \end{pmatrix}$$

 $A\mathcal{R} = \bar{U}$



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Mean value of
$$\widetilde{U}_i^n$$
 on a K_j
$$U_i^n + \sum_{1 \le |\boldsymbol{\alpha}| \le d} \mathcal{R}_{\boldsymbol{\alpha}} \left(\bar{\mathbf{x}}_j^{\boldsymbol{\alpha}} \right)$$

Packing in a Matrix-vector form

$$\begin{pmatrix} \bar{\mathbf{x}}_{1}^{(1,0,0)} & \bar{\mathbf{x}}_{1}^{(0,1,0)} & \bar{\mathbf{x}}_{1}^{(0,0,1)} & \cdots & \bar{\mathbf{x}}_{1}^{(0,0,d)} \\ \bar{\mathbf{x}}_{2}^{(1,0,0)} & \bar{\mathbf{x}}_{2}^{(0,1,0)} & \bar{\mathbf{x}}_{2}^{(0,0,1)} & \cdots & \bar{\mathbf{x}}_{2}^{(0,0,d)} \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ \bar{\mathbf{x}}_{N}^{(1,0,0)} & \bar{\mathbf{x}}_{N}^{(0,1,0)} & \bar{\mathbf{x}}_{N}^{(0,0,1)} & \cdots & \bar{\mathbf{x}}_{N}^{(0,0,d)} \end{pmatrix} \begin{pmatrix} \mathcal{R}_{(1,0,0)} \\ \mathcal{R}_{(0,1,0)} \\ \mathcal{R}_{(0,0,1)} \\ \vdots \\ \mathcal{R}_{(0,0,d)} \end{pmatrix} = \begin{pmatrix} U_{1}^{n} - U_{i}^{n} \\ U_{2}^{n} - U_{i}^{n} \\ \vdots \\ U_{N}^{n} - U_{i}^{n} \end{pmatrix}$$

 $A\mathcal{R}=\bar{U}$



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Idea for resolution

 $A\mathcal{R} = \bar{U} \Leftrightarrow A^t A \mathcal{R} = A^t \bar{U} \Leftrightarrow \mathcal{R} = (A^t A)^{-1} A^t \bar{U} \Leftrightarrow \mathcal{R} = A^\dagger \bar{U}$

How to get A^{\dagger}

- Using a QR decomposition of A: * A = QR, with $Q^t = Q^{-1}$ and R triangular superior $A^{\dagger} = ((QR)^t (QR))^{-1} A^t \Rightarrow A^{\dagger} = (R^t R)^{-1} A^t$
- Other possibilities: SVD, etc.
- Pre-processing: pseudo-inverse matrices are stored



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- Using a QR decomposition of A: * A = QR, with $Q^t = Q^{-1}$ and R triangular superior $A^{\dagger} = ((QR)^t (QR))^{-1} A^t \Rightarrow A^{\dagger} = (R^t R)^{-1} A^t$
- Other possibilities: SVD, etc.
- Pre-processing: pseudo-inverse matrices are stored





Runge-Kutta 3 TVD

Higher-order discretization in time

• The spatial high-order Finite Volume scheme

$$U_h^{n+1} = U_h^n + \Delta t \mathcal{H}^R(U_h^n), \quad U_h^n = \sum_{i, \text{ cells}} U_i^n \mathbb{1}_{\Omega_i}$$

Abstract form that includes relax, step and energies correction.

• The 3^{rd} -order Runge–Kutta time discretization

$$\begin{array}{lll} U_h^{(1)} &=& U_h^n + \Delta t \ \mathcal{H}^R \left(U_h^n \right) \\ U_h^{(2)} &=& U_h^{(1)} + \Delta t \ \mathcal{H}^R (U_h^{(1)}) \\ U_h^{(3)} &=& \bar{U}_h^{(2)} + \Delta t \ \mathcal{H}^R (\bar{U}_h^{(2)}) \end{array} \right\} \implies U_h^{n+1} = \frac{U_h^n + 2U_h^{(3)}}{3}$$

where
$$\bar{U}_{h}^{(2)} = \frac{3U_{h}^{n} + U_{h}^{(2)}}{4}$$

\implies MOOD is performed on each substep!



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HLLC RIEMANN SOLVER

Waves speeds estimates

$$S_L = \min(u_L - c_L, u_R - c_R),$$

$$S_R = \max(u_L + c_L, u_R + c_R),$$

$$S^* = \frac{p_R - p_L + (\rho u)_L (S_L - u_L) - (\rho u)_R (S_R - u_R)}{\rho_L (S_L - u_L) - \rho_R (S_R - u_R)}.$$

The HLLC is given by

$$\mathbb{F}_{HLLC}(U_L, U_R) = \begin{cases} F_L & \text{if} & 0 \le S_L, \\ F_L + S_L(U_L^{\star} - U_L) & \text{if} & S_L \le 0 \le S^{\star}, \\ F_R + S_R(U_R^{\star} - U_R) & \text{if} & S^{\star} \le 0 \le S_R, \\ F_R & \text{if} & 0 \ge S_R, \end{cases}$$

where for $K = \{L, R\}$, F_K is the physical flux and

$$U_{K}^{\star} = \begin{pmatrix} S_{K} - u_{K} \\ S_{K} - S^{\star} \end{pmatrix} \begin{pmatrix} (\alpha_{1}\rho_{1})_{K} \\ (\alpha_{2}\rho_{2})_{K} \\ \rho_{K}S^{\star} \\ \rho_{K}E_{K} + \rho_{K}(S^{\star} - u_{K}) \left(S^{\star} + \frac{p_{K}}{\rho_{K}(S_{K} - u_{K})}\right) \end{pmatrix}$$

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HLLC RIEMANN SOLVER

Additional variables

Volume fractions (constant along trajectories)

$$(\alpha_k)_K^\star = (\alpha_k)_K$$

Material densities

$$(\rho_k)_K^{\star} = (\rho_k)_K \left(\frac{S_K - u_K}{S_K - S^{\star}}\right)$$

Material pressures

$$(p_k)_K^{\star} = \left((p_k)_K + \pi_k \right) \frac{(\gamma_k - 1)(\rho_k)_K - (\gamma_k + 1)(\rho_k)_K^{\star}}{(\gamma_k - 1)(\rho_k)_K^{\star} - (\gamma_k + 1)(\rho_k)_K} - \pi_k$$

Material internal energies

$$(e_k)_K^{\star} = (e_k)_K^{\star} \left((\rho_k)_K^{\star}, (p_k)_K^{\star} \right)$$



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2-material Sod - 200 Cells - Solution



$t_{end} = 0.2s$	ρ	u	р	γ	π	Boundary Cond.
Left State on $[0.0; 0.5]$	1	0	2	2.0	0	Reflective at x=0.0
Right State on $[0.5; 1.0]$	0.125	0	1	1.4	0	Reflective at $x=1.0$

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2-material Sod - 200 Cells - Solution



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2-material Sod - 200 cells - Convergence



		MU	JSCL	M	IOOD-P1	MOOD-P3			
	1600 cells	Rel.Err.	CPU time	Rel.Err.	CPU time	Rel.Err.	CPU time		
	Density	1	10s.	1/1.2	15s. (14.4s.)	1/2.2	27s. (19.6s.)		
	Sp.Int.E.	1	10s.	1/1.5	15s. (22.5s.)	1/2.7	27s. (48.4s.)		
Estimated time for MUSCL to reach same error assuming a perfect order 1									

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AIR BUBBLE - INITIALIZATION

• S.Kokh, F.Lagoutière. "An anti-diffusive numerical scheme for the simulation of interfaces between compressible fluids by means of a five-equation model"



- Final time = $800.0 \ \mu s$
- Wall boundary conditions on top/bottom



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AIR BUBBLE - 600x300 - VOLUME FRACTION

MOOD-P1 MOOD-P3



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AIR BUBBLE - 600x300 - DENSITY

MOOD-P1 MOOD-P3



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